FTR - Flight Test Report

Dieser Prüfbericht darf ohne schriftliche Zustimmung der EAPR nicht, auch nicht auszugsweise, vervielfältigt werden

Manufacturer	SKYWALK	Type testing No.	EAPR-GS-0511/16
	Skywalk GmbH & Co.KG Windeckstr. 4 D-83250 Maquartstein	serial number	Proto
Model	Poison X-Alps S	Location	Gardasee
		Location	Gardasee



Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing 16.03.2016	Minimum take off weight 85 kg	Maximum take off weight 105 kg			
Testpilot	Johannes Tschofen	Anselm Rauh			
Harness	EAPR Equipment	EAPR light			
Pilot's take off weight	85 kg	106 kg			





Test-criteria		Minimum take off weight	Evaluation	Maximum take off weight	Evaluation
1. Inflation / take-off - 4.4.1					
Rising behavior		Easy rising, some pilot correction is required	В	Easy rising, some pilot correction is required	В
Special take off technique required		No	Α	No	Α
2. Landing - 4.4.2					
Special landing technique required		No	Α	No	Α
3. Speeds in straight flight - 4.4.3		1			
Trim speed more than 30km/h		Yes	A	Yes	Α
Speed range using the controls larger than 10km/h		Yes	A	Yes	A
Minimum speed		25 km/h to 30 km/h	В	25 km/h to 30 km/h	В
4. Control movement - 4.4.4					
Max. weight in flight up to 80kg			-		-
Max. weight in flight 80 to 100kg			-		-
Max. weight in flight greater than 100kg		Increasing 35cm - 50cm	D	Increasing 35cm - 50cm	D
5. Pitch stability exiting accelerated flight -	1.4.5	1			
Dive forward angle on exit		Dive forward less than 30°	Α	Dive forward less than 30°	Α
Collapse occurs		No	Α	No	Α
6. Pitch stability operating controls during a	ccelerated	flight - 4.4.6			
Collapse occurs		No	Α	No	Α
7. Roll stability and damping - 4.4.7					
Oscillations		Reducing	А	Reducing	Α
8. Stability in gentle spirals - 4.4.8					
Tendency to return to straight flight		Spontaneous exit	A	Spontaneous exit	Α
9. Behaviour exiting a fully developed spiral	dive - 4.4.			·	
Initial response of glider (first 180°)	tial response of glider (first 180°)		В	Immediate reduction of rate in turn	Α
Tendency to return to straight flight		Spontaneous exit	Α	Spontaneous exit	Α
Turn angle to recover normal flight		1080° to 1440°, spontaneous recovery	С	Less than 720°, spontaneous recovery	Α
10. Symmetric front collapse - 4.4.10					
Folding lines used		Yes	D	Yes	D
Entry	30%	Rocking back less than 45°	Α	Rocking back less than 45°	Α
Recovery	peeds ~ 3	Spontaneous in 3 to 5 sec	В	Spontaneous in 3 to 5 sec	В
Dive forward angle on exit	- i-	30° - 60° Entering a turn of less than 90°	В	30° - 60° Entering a turn of 90° to 180°	C
Cascade occurs		No	A	No Rocking back less than 45°	A
Entry Recovery		Rocking back less than 45° Spontaneous in 3 to 5 sec	В	Spontaneous in 3 to 5 sec	A B
Dive forward angle on exit	peeds	30° - 60° Entering a turn of 90° to 180°	С	30° - 60° Entering a turn of 90° to 180°	С
Cascade occurs	tř	No Entering a turn of 30 to 100	A	No	A
Entry	%09	Rocking back greater than 45°	C	Rocking back greater than 45°	C
Recovery		Recovery through pilot action in less than a further 3 sec	er D	Recovery through pilot action in less than a further 3 sec	r D
Dive forward angle on exit Cascade occurs	accelerated>	30° - 60° Entering a turn of 90° to 180° No	C A	30° - 60° Entering a turn of 90° to 180° No	C A
11. Exiting deep stall (parachutal stall) - 4.4.	.11				
Deep stall achieved		Yes		Yes	
Recovery		Spontaneous in 3 to 5 sec		Spontaneous in 3 to 5 sec	С
·		30° - 60°	В	0° - 30°	^
Dive forward angle on exit Change of course		Changing course 45° or more	C	Changing course less than 45°	A
Change of course Cascade occurs		onanging boulde to or more		onanging Jourse 1033 triair 40	

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12. High angle of attack recovery - 4.4.12									
Recovery			3 to 5 sec		С	Spontaneous in	1 3 to 5 sec		С
Cascade occurs	<u> </u>				Α	No			A
13. Recovery from a developed full stall - 4.4.	13								
Dive forward angle on exit		30° - 60° No collapse			В	30° - 60° No collapse			B A
Collapse Cascade occurs (other than collapse)		No collapse No			A	No collapse			A
Rocking backward		Less than 45°			Α	Less than 45°			Α
Line tension		Most lines tight			Α	Most lines tight			А
14. Asymmetric collapse (trim speed) - 4.4.14 Folding lines used		Yes			D	Yes			D
Change of course until re-inflation		< 90°	Dive or roll angle	0° - 15°	A	< 90°	Dive or roll angle	15° - 45°	A
-	trim speed, max 50% collapse			L					
Re-inflation behavior	rim speed, c 50% colla	Inflates in less than 3 sec from start of pilot action			С	Spontaneous re	Α		
Total change of course Collapse on the opposite side occurs	rim s	Less than 360° No		A	Less than 360°			A	
Twist occurs	m t	No		A	No			A	
Cascade occurs		No		Α	No			Α	
Change of course until re-inflation	Se	90° - 180°	Dive or roll angle	15° - 45°	В	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	trim speed, max 75% collapse	Inflates in less t	han 3 sec from sta	art of pilot action	С	Spontaneous re	e-inflation		Α
Total change of course	n spe	Less than 360°		Α	Less than 360°			Α	
Collapse on the opposite side occurs	ax 7	No		Α	No			Α	
Twist occurs Cascade occurs	Ε	No No			A	No No			A A
		1				<u> </u>			
Change of course until re-inflation	osc.	< 90°	Dive or roll angle	15° - 45°	Α	90° - 180°	Dive or roll angle	15° - 45°	В
Re-inflation behavior	accelerated, max 50% collapse	Inflates in less to	han 3 sec from sta	art of pilot action	С	Spontaneous re	e-inflation		Α
Total change of course	celer :0% c	Less than 360°			Α	Less than 360°	·		Α
Collapse on the opposite side occurs	acc lax 5	No			Α	No			A
Twist occurs Cascade occurs	Ε	No No			A	No No			A
Change of course until re-inflation		90° - 180°	Dive or roll angle	15° - 45°	В	90° - 180°	Dive or roll angle	45° - 60°	C
-	accelerated, max 75% collapse								
Re-inflation behavior	accelerated ıx 75% colla _l		han 3 sec from sta	art of pilot action	С		than 3 sec from sta	rt of pilot action	С
Total change of course Collapse on the opposite side occurs	locel (75%	Less than 360° No			A	Less than 360° No No			A
Twist occurs	m ax	No			A				A
Cascade occurs		No			Α	No			Α
15. Directional control with a maintained asym	metric col					L. V.			Α
Able to keep course straight		Yes		A	Yes				
180° turn away from the collapsed side possible in	rurn away from the collapsed side possible in 10 sec		Yes		Α	Yes			Α
Amount of control range between turn and stall or spin		More than 50% of the symmetric control travel		Α	A More than 50% of the symmetric control travel			Α	
16. Trim speed spin tendency - 4.4.16		•							
Spin occurs		No			Α	No			Α
17. Low speed spin tendency - 4.4.17		T.N.a.							
Spin occurs 18. Recovery from a developed spin - 4.4.18		No			Α	No			Α
			. 000 1 . 1000		С		'- 000 to 1000		С
Spin rotation angle after release		Stops spinning in 90° to 180°			Stops spinning in 90° to 180°				
Cascade occurs 19. B-line-stall - 4.4.19		No		Α	No			Α	
Change of course before release		1			NA				NA
Behaviour before release					NA				NA
			-		NIA				NIA
Recovery		-			NA				NA
Dive forward angle on exit Cascade occurs				NA NA				NA NA	
20. Big ears - 4.4.20									
Entry procedure		Special device required		Α	Special device	required		Α	
Behaviour during big ears		Unstable flight		С	Stable flight			А	
Recovery		Recovery through pilot action in less than a further		В	Recovery through pilot action in less than a further			В	
Dive forward angle on exit		3 sec 0° - 30°		A	3 sec 0° bis 30°			A	
21. Big Ears in accelerated flight - 4.4.21		,							
Entry procedure		Special device required		Α	Special device required			Α	
Behaviour during big ears		Unstable flight		C	Stable flight			A	
Recovery		Recovery through pilot action in less than a further		В	Recovery through pilot action in less than a further			В	
Dive forward angle on exit		3 sec 0° - 30°		A	3 sec 0° bis 30°			A	
Behaviour immediately after releasing the accelarator while		Unstable flight		C	Stable flight			A	
maintaining big ears	1 / 22	J				1			,,
23. Alternative means of directional control - 4	1.4.22	1				ı			
180° turn achievable in 20 sec		Yes		Α	Yes			Α	
Stall or spin occurs	ation dec	No	'e manual 4.4	22	Α	No			Α
23. Any other flight procedure and/or configure Procedure works as descibed	ation desc	indea in the user	s manuai - 4.4.	23	NA				NA
Procedure works as described Procedure suitable for novice pilots		<u> </u>		NA				NA	
Cascade occurs				NA				NA	
24. Remarks of testpilot:									
Í		L				L			