## FTR - Flight Test Report

Manufacturer		Type testing No.	EAPR-GS-0544/16	
	Hochriess traße 1 D-83128 Flints bach	serial number		
Model			Schruns	
Comment		Location	Gerlitzen, Ossiacher See	



Rev. 2.3 - 26.11.2014 EAPR GmbH - Marktstr. 11 D-87730 Bad Grönenbach - Germany

Date of testing	22.08.2016	Minimum take 90 kg		Maximum take off weight 115 kg		
Testpilot		Hannes Tschofen		Anselm Rauh		
Harness		EAPR Equipment		EAPR schwer	No.	
Pilot's take off weight	1	115	kg 🚮	115	kg	

Classification В



Test-criteria		Minimum take off weight		Maximum take off weight	Evaluatio	
1. Inflation / take-off - 4.4.1						
Rising behavior		no pilot correction required		no pilot correction required	А	
Special take off technique required			Α	No	А	
2. Landing - 4.4.2	No		7.	110		
Special landing technique required	No		A	No	l A	
3. Speeds in straight flight - 4.4.3	140		A	140	A	
	Yes			I V.		
Trim speed more than 30km/h			Α	Yes	A	
Speed range using the controls larger than 10km/h			Α	Yes	Α	
Minimum speed		than 25 km/h	Α	Less than 25 km/h	Α	
4. Control movement - 4.4.4	•					
Max. weight in flight up to 80kg			-		-	
Max. weight in flight 80 to 100kg					-	
Max. weight in flight greater than 100kg	Increa	sing >65 cm	Α	Increasing >65 cm	Α	
5. Pitch stability exiting accelerated flight - 4.4.5						
Dive forward angle on exit		Dive forward less than 30°		Dive forward less than 30°	А	
Collapse occurs			A	No	A	
6. Pitch stability operating controls during accele	rated flight - 4	.4.6				
Collapse occurs	No		Α	No	Α	
7. Roll stability and damping - 4.4.7			, , ,			
Oscillations	Reduc	ing	A	Reducing	l A	
	rieduc	sing	A	rieddcing	A	
8. Stability in gentle spirals - 4.4.8	10.					
Tendency to return to straight flight		aneous exit	Α	Spontaneous exit	Α	
9. Behaviour exiting a fully developed spiral dive						
Initial response of glider (first 180°)		mediate reaction	В	No immediate reaction	B	
Tendency to return to straight flight		aneous exit	A	Spontaneous exit		
Turn angle to recover normal flight		o 1080°, spontaneous recovery	В	720° to 1080°, spontaneous recovery	В	
10. Symmetric front collapse - 4.4.10						
Folding lines used	No			No		
Entry	Rocki	Rocking back less than 45°  A Rocking back less than 45°		Rocking back less than 45°	Α	
Recovery		aneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	0° - 30	)° Keeping course	Α	0° - 30° Keeping course	Α	
Cascade occurs	E No		Α	No	Α	
Entry	Rocki	ng back less than 45°	Α	Rocking back less than 45°	Α	
Recovery	Spont	aneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	€ 0 00	o° Entering a turn of less than 90°	Α	0° - 30° Keeping course	Α	
Cascade occurs	No No		Α	No	Α	
Entry	Rocki	ng back less than 45°	Α	Rocking back less than 45°	Α	
Recovery		aneous in less than 3 sec	Α	Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	e 0° - 30	0° Entering a turn of less than 90°	Α	0° - 30° Entering a turn of less than 90°	Α	
Cascade occurs	<sup>8</sup> No		Α	No	Α	
11. Exiting deep stall (parachutal stall) - 4.4.11						
Deep stall achieved	Yes			Yes		
Recovery	Spont	Spontaneous in less than 3 sec		Spontaneous in less than 3 sec	Α	
Dive forward angle on exit	0° - 30	0° - 30° A 0° - 30°		0° - 30°	Α	
Change of course		Changing course less than 45°		Changing course less than 45°	Α	
Cascade occurs				No		

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Recovery Cascade occurs									
	12. High angle of attack recovery - 4.4.12 Recovery			Spontaneous in less than 3 sec A			Spontaneous in less than 3 sec		
10. Deservent from a developed full stell. 4.4	•		No			No			Α
13. Recovery from a developed full stall - 4.4.13					Α	1.72			
Dive forward angle on exit		0° - 30°			Α	30° - 60°			В
Collapse Cascade occurs (other than collapse)		No collapse No			A	No collapse No			A A
Rocking backward		Less than 45°			A	Less than 45°			A
Line tension		Most lines tight			Α	Most lines tight			Α
14. Asymmetric collapse (trim speed) - 4.4.14		Late				L M.			
Folding lines used		No				No	I		
Change of course until re-inflation	bse	< 90°	Dive or roll angle	0° - 15°	Α	< 90°	Dive or roll angle	15° - 45°	Α
Re-inflation behavior	trim speed, max 50% collapse	Spontaneous re-inf	lation		Α	Spontaneous re	-inflation		Α
Total change of course	trim speed x 50% colla	Less than 360°			Α	Less than 360° No No			Α
Collapse on the opposite side occurs Twist occurs	nax (	No No		A	A A				
Cascade occurs		No			A	No			
Change of course until re-inflation	trim speed, max 75% collapse	< 90°	Dive or roll angle	15° - 45°	Α	< 90°	Dive or roll angle	15° - 45°	Α
		<u> </u>				_	1	l	
Re-inflation behavior	beec col	Spontaneous re-inf	lation		Α	Spontaneous re	-inflation		Α
Total change of course Collapse on the opposite side occurs	trim speed x 75% colls	Less than 360°		A	Less than 360° No			A A	
Twist occurs	m ax	No		A	No No			A	
Cascade occurs		No		Α	No			Α	
Change of course until re-inflation		< 90°	Dive or roll angle	0° - 15°	Α	< 90°	Dive or roll angle	15° - 45°	Α
	d, apse			J 10			_	.5 40	
Re-inflation behavior	colle	Spontaneous re-inf	lation		Α	Spontaneous re	-inflation		Α
Total change of course	accelerated, max 50% collapse	Less than 360°			A	Less than 360°			A
Collapse on the opposite side occurs Twist occurs	ас	No No			A	No No			A
Cascade occurs		No			Ä	No			A
Change of course until re-inflation	Φ	90° - 180°	Dive or roll angle	15° - 45°	В	90° - 180°	Dive or roll angle	15° - 45°	В
De la Caller de la lace la caller	accelerated, max 75% collapse	0	1.0	<u>I</u>	•	0	1.0.00	l	
Re-inflation behavior	erate % col	Spontaneous re-inf	ation		Α	Spontaneous re	-inflation		Α
Total change of course  Collapse on the opposite side occurs	accelerated x 75% collap	Less than 360° No No			A	Less than 360° No			A
Twist occurs	ma <sub>0</sub>				A	No			A
Cascade occurs		No			Α	No			Α
15. Directional control with a maintained asymphotography Able to keep course straight	imetric co	Yes			A	Yes			Λ.
	- 10	Yes							A
180° turn away from the collapsed side possible in 10 sec		165			Α	Yes			Α
Amount of control range between turn and stall or spin		More than 50% of the symmetric control travel			Α	More than 50%	of the symmetric	control travel	Α
16. Trim speed spin tendency - 4.4.16		•				•			
Spin occurs		No			Α	No			Α
17. Low speed spin tendency - 4.4.17						1			
Spin occurs		No			Α	No			Α
18. Recovery from a developed spin - 4.4.18		1				ı			
Spin rotation angle after release		Stops spinning in less than 90°			Α	Stops spinning in less than 90°			Α
Cascade occurs		No			Α	No	А		
19. B-line-stall - 4.4.19		I Chanaina saumas la	the 450		Δ.	I Chansian anum	a lana than AEO		Δ.
Change of course before release		Changing course less than 45°			Α	Changing course less than 45°			A
Behaviour before release		Remains stable with straight span		Α	Remains stable with straight span			Α	
Recovery	Recovery		Spontaneous in less than 3 sec			Spontaneous in less than 3 sec			Α
Dive forward angle on exit		0° - 30°			Α	0° - 30°			Α
Cascade occurs		No			Α	No			Α
20. Big ears - 4.4.20									
Entry procedure		Special device required			Α	Special device required			Α
Behaviour during big ears		Stable flight			Α	Stable flight			Α
Recovery		Spontaneous in less than 3 sec			Α	Spontaneous in less than 3 sec			Α
Dive forward angle on exit		0° - 30°			Α	0° bis 30°			Α
21. Big Ears in accelerated flight - 4.4.21									
Entry procedure		Special device required		Α	Special device required			Α	
Behaviour during big ears		Stable flight		Α	Stable flight			Α	
Recovery		Spontaneous in 3 to 5 sec		Α	Spontaneous in less than 3 sec			Α	
Recovery	·		0° - 30°		A	0° bis 30°			A
•	Behaviour immediately after releasing the accelarator while		Stable flight		A	Stable flight			A
Dive forward angle on exit Behaviour immediately after releasing the accelar	rator while				7.	Stable Hight			,,
Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears		Stable Hight							
Dive forward angle on exit Behaviour immediately after releasing the accelar		Stable Hight							
Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears		Yes Yes			А	Yes			А
Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears  23. Alternative means of directional control -		-			A A	Yes No			A A
Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears  23. Alternative means of directional control  180° turn achievable in 20 sec  Stall or spin occurs  23. Any other flight procedure and/or configur	4.4.22	Yes No	manual - 4.4.	23	А				A
Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears  23. Alternative means of directional control - 180° turn achievable in 20 sec  Stall or spin occurs  23. Any other flight procedure and/or configur Procedure works as descibed	4.4.22	Yes No	manual - 4.4.	23	A				A NA
Dive forward angle on exit Behaviour immediately after releasing the accelar maintaining big ears  23. Alternative means of directional control  180° turn achievable in 20 sec  Stall or spin occurs  23. Any other flight procedure and/or configur	4.4.22	Yes No	manual - 4.4.	23	A				A

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